

# From reduced-order modeling to scientific machine learning

How computational science is enabling the  
design of next-generation aerospace systems

Professor Karen E. Willcox  
SDM Lecture | January 5, 2022



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FOR COMPUTATIONAL ENGINEERING & SCIENCES



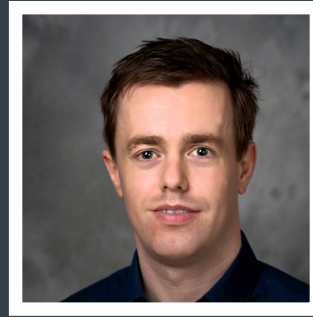
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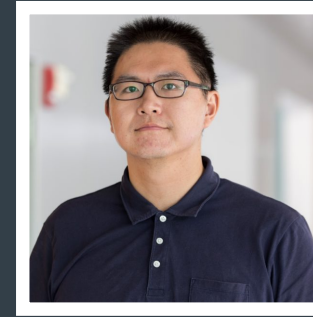
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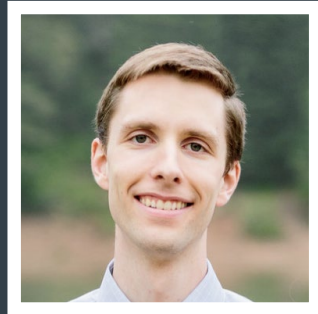
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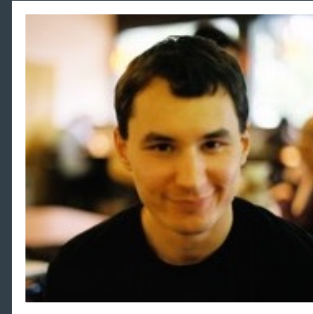
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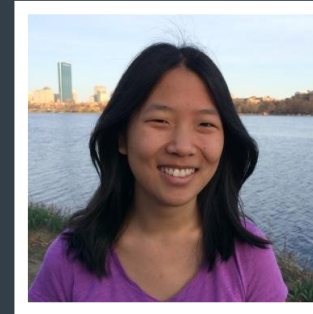
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# Outline

## 1 Learning from data

The imperative of physics-based modeling & inverse theory

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## 2 Reduced-order modeling

A critical enabler for accelerating predictive computations in support of engineering design

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## 3 Operator Inference

Combining model reduction & machine learning to learn predictive reduced-order models

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## 4 Outlook

● **Forward simulations**

The backbone of engineering analysis

● **Uncertainty quantification**

Design for reliability & robustness

● **Optimization**

Optimal design, inverse problems, control, parameter estimation

● **Scientific machine learning**

Blending physics modeling & data-driven learning



**Computational science has been enabling engineering design for six decades**



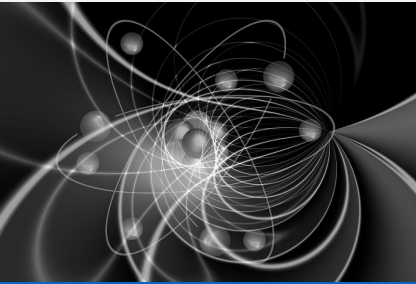
# Scientific Machine Learning

What are the opportunities and challenges of machine learning in complex applications across science, engineering, and medicine?

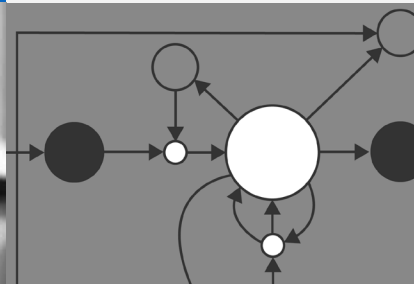
Embed domain knowledge



Respect physical constraints



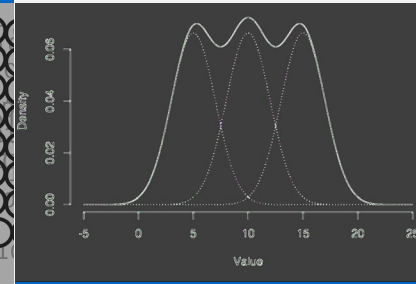
Bring interpretability to results



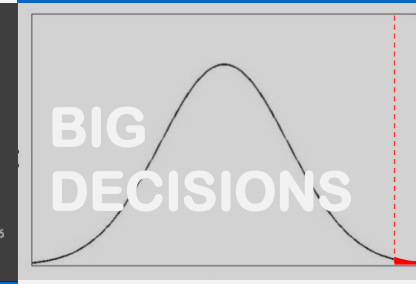
Integrate sparse, heterogeneous, noisy & incomplete data



Make predictions with quantified uncertainties



Support high-consequence decisions



THIS IS YOUR MACHINE LEARNING SYSTEM?

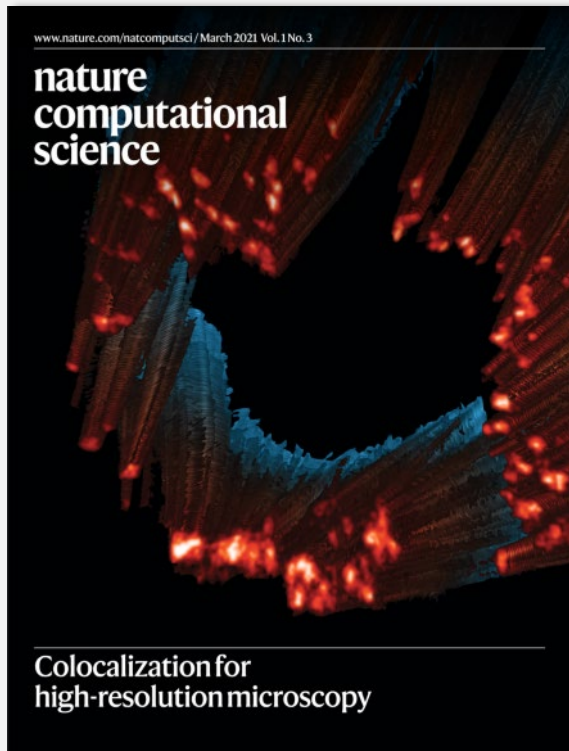
YUP! YOU POUR THE DATA INTO THIS BIG PILE OF LINEAR ALGEBRA, THEN COLLECT THE ANSWERS ON THE OTHER SIDE.

WHAT IF THE ANSWERS ARE WRONG?

JUST STIR THE PILE UNTIL THEY START LOOKING RIGHT.



<https://xkcd.com/1838/>



comment



## The imperative of physics-based modeling and inverse theory in computational science

To best learn from data about large-scale complex systems, physics-based models representing the laws of nature must be integrated into the learning process. Inverse theory provides a crucial perspective for addressing the challenges of ill-posedness, uncertainty, nonlinearity and under-sampling.

Karen E. Willcox, Omar Ghattas and Patrick Heimbach

The notions of ‘artificial intelligence (AI) for science’ and ‘scientific machine learning’ (SciML) are gaining widespread attention in the scientific community. These initiatives target development and adoption of AI approaches in scientific and engineering fields with the goal of accelerating research and development breakthroughs in energy, basic science, engineering, medicine and national security. For the past six decades, these fields have been advanced through the synergistic and principled use of

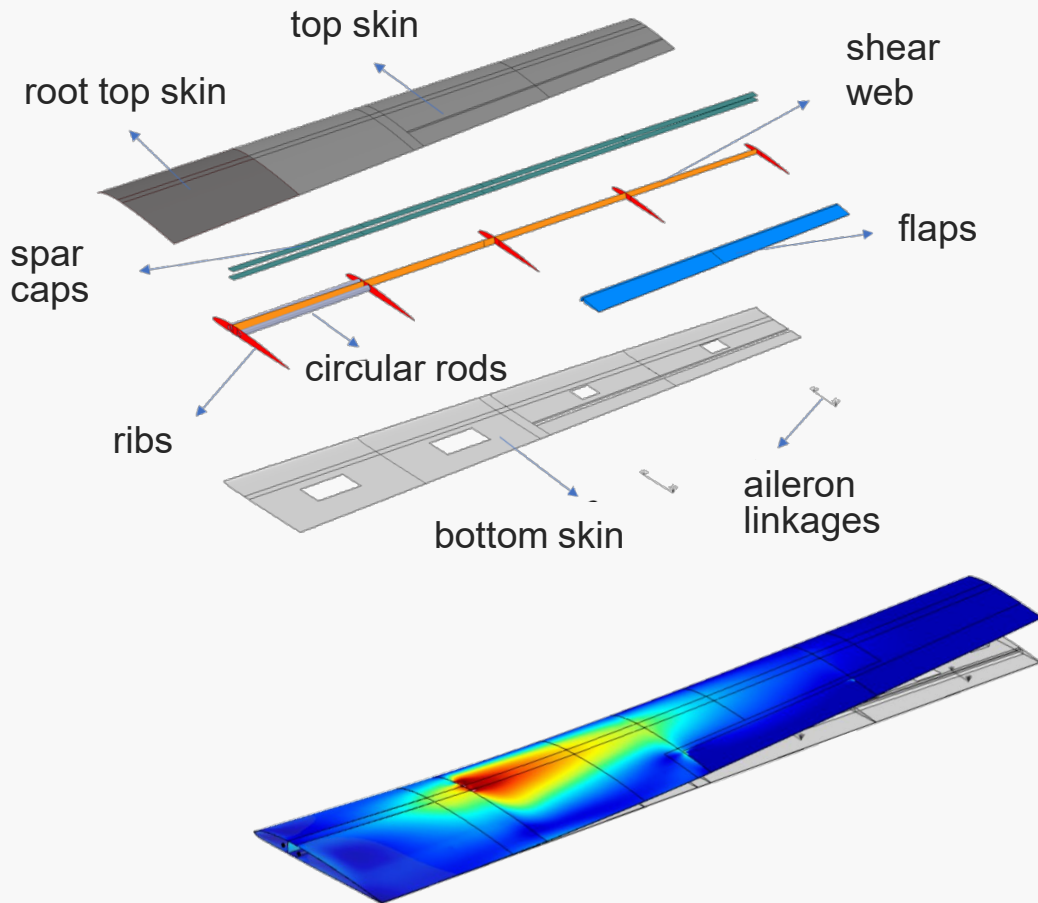
geological processes evolve. Physics-based models typically encode knowledge in the form of conservation and constitutive laws, often based on decades if not centuries of theoretical development and experimental validation. These laws often manifest as systems of differential equations that are solved numerically with high-performance computing (HPC).

In his famous 1960 article, Eugene Wigner wrote about “The unreasonable effectiveness of mathematics in the natural sciences”, pointing to “the ‘laws of nature’

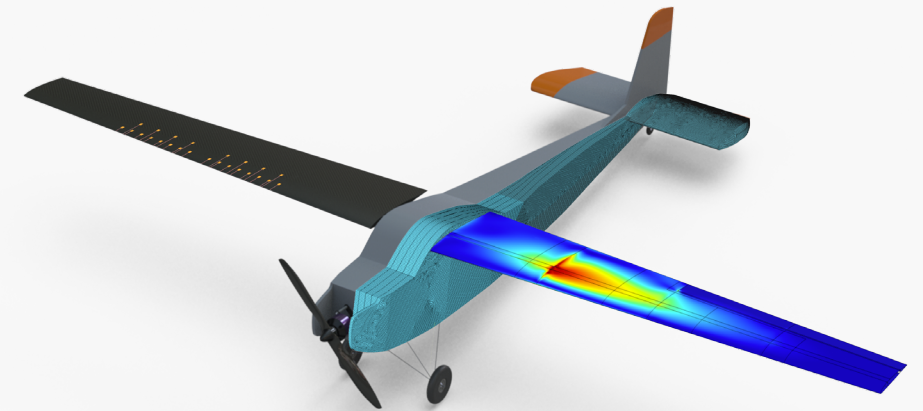
constraints, purely data-driven approaches are unlikely to be predictive, no matter how expressive the underlying representation. Even when physical models are not well-established (such as for many biological processes, in constitutive laws for complex materials, or in subgrid scale models for unresolved physics), we know that certain universal properties and relationships must hold, such as conservation properties, material frame indifference, objectivity, symmetries, or other invariants. The learning-from-data



# PHYSICS-BASED MODELS are POWERFUL and bring PREDICTIVE CAPABILITIES



but they can be  
**COMPUTATIONALLY  
EXPENSIVE**



# COMPUTATIONAL SCIENCE enables AEROSPACE DESIGN

1

## Learning from data

The imperative of physics-based modeling & inverse theory

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2

## Reduced-order modeling

A critical enabler for accelerating predictive computations in support of engineering design

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3

## Operator Inference

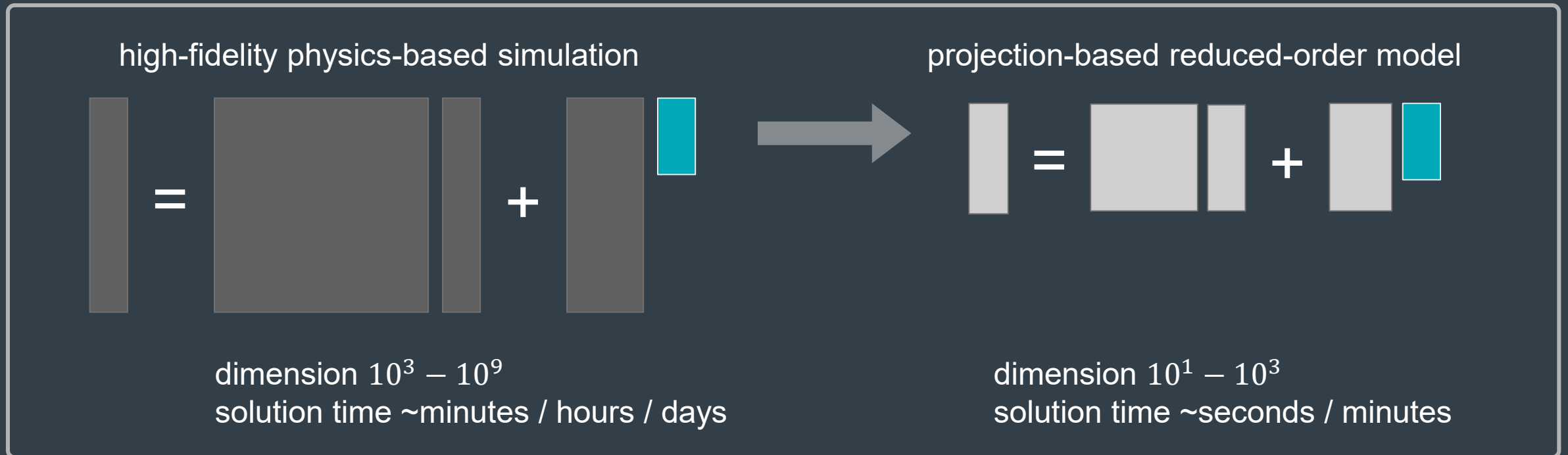
Combining model reduction & machine learning to learn predictive reduced-order models

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4

## Outlook

# Reduced-order models are critical enablers for data-driven learning & engineering design



- 1 Train:** Solve PDEs to generate training data (snapshots)
- 2 Identify structure:** Compute a low-dimensional basis
- 3 Reduce:** Project PDE model onto the low-dimensional subspace



# Technical Notes

TECHNICAL NOTES are short manuscripts describing new developments or important results of a preliminary nature. These Notes cannot exceed 6 manuscript pages and 3 figures; a page of text may be substituted for a figure and vice versa. After informal review by the editors, they may be published within a few months of the date of receipt. Style requirements are the same as for regular contributions (see inside back cover).

## An Approximate Analysis Technique for Design Calculations

R. L. FOX\* and H. MIURA†  
Case Western Reserve University, Cleveland, Ohio

### 1 Introduction

IN the design of complex structures, it is often necessary or desirable to employ approximations in the analysis to reduce computational cost and required computer storage. In automated or computer assisted design applications, it is often necessary to analyze a considerable number of designs, and it is the computational cost of these analyses that inhibits applications in many cases. Although no specific optimization problem is formulated in this Note, the method proposed is particularly applicable to such problems.

In this Note, a simple method is proposed with which one can obtain approximate results for analyses of modified designs based upon a limited number of exact analysis results. The idea of this method is based on the practically experienced fact that the number of design variables are usually far smaller than the degrees of freedom of the system, and the further observation that often large numbers of analysis degrees of freedom are dictated by the topology of the design rather than by the expected complexity of its behavior.

Some encouraging numerical examples, computed for space truss structures, are presented.

### 2 Approximate Method

In the static analysis of a structure using the displacement method, the system is expressed in the form

$$K X = P \quad (1)$$

where  $K$  is the stiffness matrix,  $X$  is the vector of displacement degrees of freedom, and  $P$  is the load vector. If the structure has  $t$  design variables ( $d_1, d_2, \dots, d_t$ ), we can consider this set as a vector  $D$  in  $t$ -dimensional space. In this design space, consider the  $r$  "basic" designs given by a set of the design vectors  $D_1, D_2, \dots, D_r$ . Corresponding to these basic design vectors, the stiffness matrices  $K_1, K_2, \dots, K_r$  are obtained, and by solving the  $r$  sets of simultaneous equations,

$$K_i X_i = P, \quad i = 1, 2, \dots, r \quad (2)$$

basic displacement vectors  $X_1, X_2, \dots, X_r$  are computed.

Table 1 Basic designs

Member	Design 1	Design 2	Design 3	Design 4	Design 5
1 ~ 18	1.0m <sup>3</sup>	0.2m <sup>3</sup>	1.0m <sup>3</sup>	1.0m <sup>3</sup>	1.0m <sup>3</sup>
19 ~ 42	1.0	1.0	0.5	1.0	1.0
43 ~ 72	2.0	2.0	2.0	0.4	2.0
73 ~ 124	0.5	0.5	0.5	0.5	0.1

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\* Associate Professor of Engineering. Member AIAA.  
† Graduate Assistant.

Consider now a new design vector  $D_n$  in the  $n$ th of the basic design vectors. The stiffness matrix  $K_n$  due to  $D_n$  can be computed as  $K_n$ , and the exact displacement vector  $X_n$  due to the external load  $P$  would ordinarily be obtained by solving a set of simultaneous equations:

$$K_n X_n = P$$

Here it is assumed that  $X_n$  can be approximated by a linear combination of basic displacement vectors  $X_i$ .

$$X_n \approx \bar{X}_n = y_1 X_1 + y_2 X_2 + \dots + y_r X_r$$

If  $X_1, X_2, \dots, X_r$  are linearly independent and  $r = n$ , the exact solution, but we are now considering a  $e < n$ . In matrix form, Eq. (4) will be expressed

$$\bar{X}_n = T Y$$

where

$$T = \begin{bmatrix} X_1 & X_2 & \dots & X_r \\ y_1 & y_2 & \dots & y_r \end{bmatrix}$$

The vector  $Y$  is then determined by solving a set of Eqs. (6), which is obtained by substituting Eq. (3) assuming  $\bar{X}_n \approx X_n$  and premultiplying  $T^T$ , i.e.,

$$T^T K_n T Y = T^T P$$

Introducing the notation

$$K_n = T^T K_n T; \quad P_n = T^T P$$

we have

$$K_n Y = P_n$$

In short, an approximate displacement vector  $\bar{X}_n$

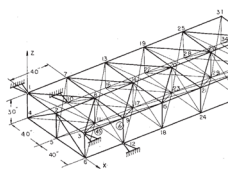


Fig. 1 Example structure.



## Reduced Basis Technique for Nonlinear Analysis of Structures

Ahmed K. Noor\* and Jeanne M. Peters†  
George Washington University Center, NASA Langley Research Center, Hampton, Va.

A reduced basis technique and a computational algorithm are presented for predicting the nonlinear static response of structures. A total Lagrangian formulation is used and the structure is discretized by using displacement finite element models. The nodal displacement vector is expressed as a linear combination of a small number of basis vectors and a Rayleigh-Ritz technique is used to approximate the finite element equations by a reduced system of nonlinear equations. The Rayleigh-Ritz approximation functions (basis vectors) are chosen to be those commonly used in the static perturbation technique namely, a nonlinear solution and a number of its path derivatives. A procedure is outlined for automatically selecting the load (or displacement) step size and monitoring the solution accuracy. The high accuracy and effectiveness of the proposed approach is demonstrated by means of numerical examples.

### Nomenclature

- $A$  = cross sectional area
- $E$  = Young's modulus of the material
- $e$  = error norm defined by Eq. (16)
- $f$  =  $[f(X, q)]$  and  $[f(\psi, q)]$  = vectors defined in Eqs. (1) and (4), respectively
- $G$  = shear modulus of the material
- $\{G(X)\}$  = vector of nonlinear terms
- $\{G(\psi)\}$  = vector of nonlinear terms of the reduced system
- $h$  = arch thickness
- $I$  = moment of inertia
- $[K]$  = linear global stiffness matrix of the structure
- $\{K\}$  = linear stiffness matrix of the reduced system
- $K_{ij}$  =  $K_{ij} + \partial G_{ij} / \partial X_j$
- $L$  = length of beam
- $l_i, l_j$  = lengths of vectors  $\{X\}$  and  $\{\psi\}$
- $N$  = normal force
- $n$  = total number of degrees-of-freedom in the finite element model
- $P$  = applied concentrated load
- $\{Q\}$  = normalized external load vector
- $\{Q\}$  = normalized load vector of the reduced system
- $q$  = load parameter
- $R$  = local radius of curvature of the arch
- $\{R\}$  = residual vector defined by Eq. (15)
- $r$  = number of basis vectors (reduced degrees-of-freedom)
- $S_{(i)}$  = current stiffness parameter corresponding to  $i$ th load (or displacement) increment
- $U$  = total strain energy

- $u, w$  = tangential (circumferential)  $a$  (radial) displacement; center line of the arch
- $\{X\}$  = vector of unknown nodal displacements
- $\{X_i\}$  ( $i = 1$  to  $r$ ) = basis vectors
- $\{X\}, \{X_i\}, \{X\}$  = derivatives of  $\{X\}$  with respect to parameter  $q$
- $\beta$  = condition number of  $[K]$
- $\{P\}$  = matrix of basis vectors
- $\theta$  = half subtended angle of the arch
- $\lambda$  = path parameter
- $\{\psi\}$  = vector of undetermined coefficients

### Introduction

LARGE deflection nonlinear analysis has been the focus of intense efforts because of the use of new lightweight materials (such as fibrous carbon aircraft and aerospace structures) and the harsh to which these structures are often subjected. progress has been made in the development of powerful finite element discretization methods improved numerical methods and program for nonlinear analysis of structures (see, for example, Ref. 1). In spite of these advances, the solutions of nonlinear structural problems require excessive computer time and, therefore, are not economical.

The large numbers of degrees-of-freedom structural systems are often discretized by their top than by the expected complexity of the behavior. been recognized and used to advantage in an optimum design and vibration analysis<sup>2,3</sup> and nonlinear analysis.<sup>4,5</sup> In the latter case a hybrid been used which combines contemporary finite classical Rayleigh-Ritz approximations, thereby the modeling versatility inherent in the finite element, and, at the same time, reducing the number of freedom through Rayleigh-Ritz approximation.

Since the effectiveness of this approach depends, to a great extent, on the approximation of the Rayleigh-Ritz approximation functions, it will be referred to herein as reduced basis vectors. 8-10, the linear bifurcation buckling modes basis vectors and only mildly nonlinear pre- considered. In contrast, Ref. 11 used the linear corrections to it as basis vectors and presented controlling the errors in nonlinear analysis.

The aforementioned choices for basis vectors to realize the full potential of the reduced basis technique, on the one hand, the generation of bifurcation

Ann. Rev. Fluid Mech. 2001.33:445-469. Downloaded from www.annualreviews.org. Access provided by University of Texas - Austin on 12/30/21. For personal use only.

## MODELING OF FLUID-STRUCTURE INTERACTION

Earl H. Dowell and Kenneth C. Hall

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**Key Words:** time linearization, nonlinear dynamics, reduced-order models, aeroelasticity

**Abstract:** The interaction of a flexible structure with a flowing fluid in which it is submerged or by which it is surrounded gives rise to a rich variety of physical phenomena with applications in many fields of engineering, for example, the stability and response of aircraft wings, the flow of blood through arteries, the response of bridges and tall buildings to winds, the vibration of turbine and compressor blades, and the oscillation of heat exchangers. To understand these phenomena we need to model both the structure and the fluid. However, in keeping with the overall theme of this volume, the emphasis here is on the fluid models. Also, the applications are largely drawn from aerospace engineering although the methods and fundamental physical phenomena have much wider applications. In the present article, we emphasize recent developments and future challenges. To provide a context for these, the article begins with a description of the various physical models for a fluid undergoing time-dependent motion, then moves to a discussion of the distinction between linear and nonlinear models, time-linearized models and their solution in either the time or frequency domains, and various methods for treating nonlinear models, including time marching, harmonic balance, and systems identification. We conclude with an extended treatment of the modal character of time-dependent flows and the construction of reduced-order models based on an expansion in terms of fluid modes. The emphasis is on the enhanced physical understanding and dramatic reductions in computational cost made possible by reduced-order models, time linearization, and methodologies drawn from dynamical system theory.

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# Our **Operator Inference** approach blends model reduction & machine learning

- ❑ A **physics-based model**  
Typically described by PDEs or ODEs
- ❑ Lens of **projection** to define the form of a  
structure-preserving low-dimensional model

Define the **structure of the reduced model**

$$\dot{\hat{\mathbf{x}}} = \hat{\mathbf{A}}\hat{\mathbf{x}} + \hat{\mathbf{B}}\mathbf{u} + \hat{\mathbf{H}}(\hat{\mathbf{x}} \otimes \hat{\mathbf{x}})$$



**Non-intrusive learning** by inferring reduced model operators  $\hat{\mathbf{A}}$ ,  $\hat{\mathbf{B}}$ ,  $\hat{\mathbf{H}}$  from simulation data

## Learning from data through the lens of a reduced-order physics-based model

# What is a physics-based model?

A representation of the **governing laws of nature** that innately embeds the concepts of **time, space, and causality**

In solving the governing equations of the system, we constrain the **predictions** to lie on the **solution manifold** defined by the laws of nature

**Example:**  
equations  
of linear  
elasticity

$\rho \frac{\partial^2 u}{\partial t^2} = \frac{\partial \sigma}{\partial x} + \frac{\partial \sigma}{\partial y} + F$	$\varepsilon = \frac{1}{2} [\nabla u + (\nabla u)^T]$	$\sigma = C : \varepsilon$	+ boundary conditions + initial conditions
equation of motion (Newton's 2 <sup>nd</sup> law)	strain-displacement equations	constitutive equations	

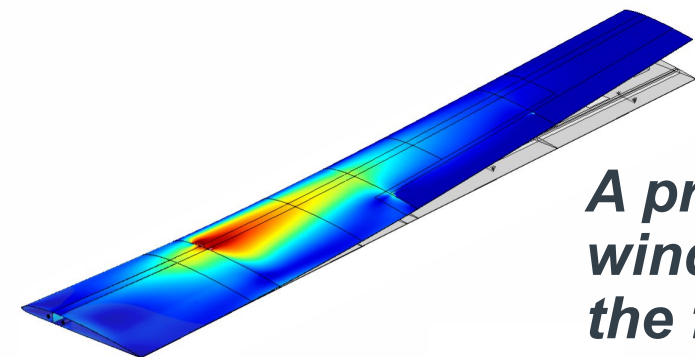
a mathematical model of how solid objects deform, relating stress  $\sigma$ , strain  $\varepsilon$ , displacement  $u$ , and loading  $F$

## The unreasonable effectiveness of physics-based models [Wigner, 1960]

**Solving a physics-based model:**

**Given** initial conditions, boundary conditions, loading conditions, and system parameters

**Compute** solution trajectories  $\sigma(x, y, t), \varepsilon(x, y, t), u(x, y, t), \dots$



***A predictive window on the future***

# What is a physics-based model?

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a mathematical model of how solid objects deform, relating stress  $\sigma$ , strain  $\varepsilon$ , displacement  $u$ , and loading  $F$

## A mathematical model solved with computational science

**Discretize:**  
Spatially discretized  
computational fluid  
dynamic (CFD) model

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u} + \mathbf{f}(\mathbf{x}, \mathbf{u})$$

discretized **state  $\mathbf{x}$**  contains physical states at  $N_z$  spatial grid points –  $N_z \sim O(10^4 - 10^6)$

$$\mathbf{x} = \begin{bmatrix} u_1 \\ u_2 \\ \vdots \\ \vdots \\ u_{N_z} \end{bmatrix}$$

e.g., nodal displacements

# Projecting a linear system

Full-order model (FOM)  
state  $\mathbf{x} \in \mathbb{R}^N$

$$\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$$



Approximate

$$\mathbf{x} \approx \mathbf{V}\mathbf{x}_r$$
$$\mathbf{V} \in \mathbb{R}^{N \times r}$$

Residual:  $N$  eqs  $\gg r$  dof

$$\mathbf{r} = \mathbf{V}\dot{\mathbf{x}}_r - \mathbf{A}\mathbf{V}\mathbf{x}_r - \mathbf{B}\mathbf{u}$$



Project

$$\mathbf{W}^\top \mathbf{r} = 0$$

(Galerkin:  $\mathbf{W} = \mathbf{V}$ )

Reduced-order model (ROM)  
state  $\mathbf{x}_r \in \mathbb{R}^r$

$$\dot{\mathbf{x}}_r = \mathbf{A}_r \mathbf{x}_r + \mathbf{B}_r \mathbf{u}$$

$$\mathbf{A}_r = \mathbf{V}^\top \mathbf{A} \mathbf{V}$$
$$\mathbf{B}_r = \mathbf{V}^\top \mathbf{B}$$



# Linear Model

**FOM:**  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{B}\mathbf{u}$



**ROM:**  $\dot{\mathbf{x}}_r = \mathbf{A}_r\mathbf{x}_r + \mathbf{B}_r\mathbf{u}$

Precompute the ROM matrices:

$$\mathbf{A}_r = \mathbf{V}^\top \mathbf{A} \mathbf{V}, \quad \mathbf{B}_r = \mathbf{V}^\top \mathbf{B}$$

# Quadratic Model

**FOM:**  $\dot{\mathbf{x}} = \mathbf{A}\mathbf{x} + \mathbf{H}(\mathbf{x} \otimes \mathbf{x}) + \mathbf{B}\mathbf{u}$



**ROM:**  $\dot{\mathbf{x}}_r = \mathbf{A}_r\mathbf{x}_r + \mathbf{H}_r(\mathbf{x}_r \otimes \mathbf{x}_r) + \mathbf{B}_r\mathbf{u}$

Precompute the ROM matrices and tensor:

$$\mathbf{H}_r = \mathbf{V}^\top \mathbf{H}(\mathbf{V} \otimes \mathbf{V})$$

**projection preserves structure  $\leftrightarrow$  structure embeds physical constraints**

# COMPUTATIONAL SCIENCE enables AEROSPACE DESIGN

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## Learning from data

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## Reduced-order modeling

A critical enabler for accelerating predictive computations in support of engineering design

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## Operator Inference

Combining model reduction & machine learning to learn predictive reduced-order models

---

4

## Outlook

# The **Operator Inference** problem

**Given** (1) a physical/natural system with known governing equations, and (2) a set of data in the form of state snapshots (experimental or simulation)

**Infer** a reduced-order model that recovers the given data and provides a predictive capability to rapidly simulate unseen conditions

$$\min_{\mathbf{O}} \|\mathbf{D}\mathbf{O} - \mathbf{R}\|$$

$\mathbf{O}$  : low-dimensional operators that define the reduced model

$\mathbf{D}$ ,  $\mathbf{R}$  : data matrix / forcing from simulation and/or experimental data

We will use:

- the **physics** to define the structured form of the model we seek
- **projection-based model reduction** to cast the inference in a reduced coordinate space and to provide error estimates in some settings
- **inverse theory** to analyze the structure of the resulting problem and treat it numerically
- **numerical linear algebra** to achieve efficient scalable algorithms

# Our **Operator Inference** approach blends model reduction & machine learning

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**Non-intrusive learning** by inferring reduced operators from simulation data [Peherstorfer & W., 2016]

snapshots generated from  
projected simulation data

low-dimensional  
operators define the  
reduced model  
as a dynamical  
system

$$\min_{\hat{\mathbf{A}}, \hat{\mathbf{B}}, \hat{\mathbf{H}}} \left\| \hat{\mathbf{X}}^T \hat{\mathbf{A}}^T + (\hat{\mathbf{X}} \otimes \hat{\mathbf{X}})^T \hat{\mathbf{H}}^T + \mathbf{U}^T \hat{\mathbf{B}}^T - \dot{\hat{\mathbf{X}}}^T \right\|$$

minimum residual  
formulation leads to  
linear least squares

- Regularization is key [McQuarrie, Huang & W., 2021]
- If data are Markovian, Operator Inference **recovers the intrusive POD ROM** [Peherstorfer, 2020]

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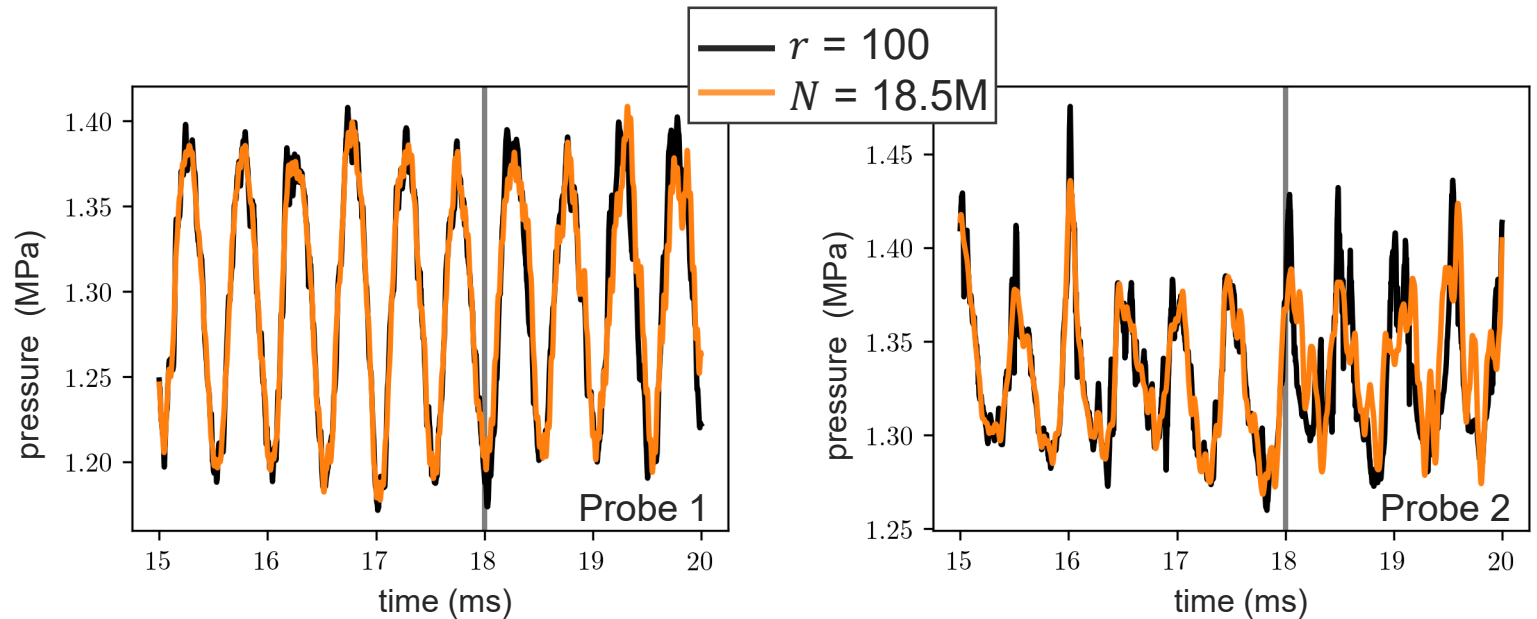
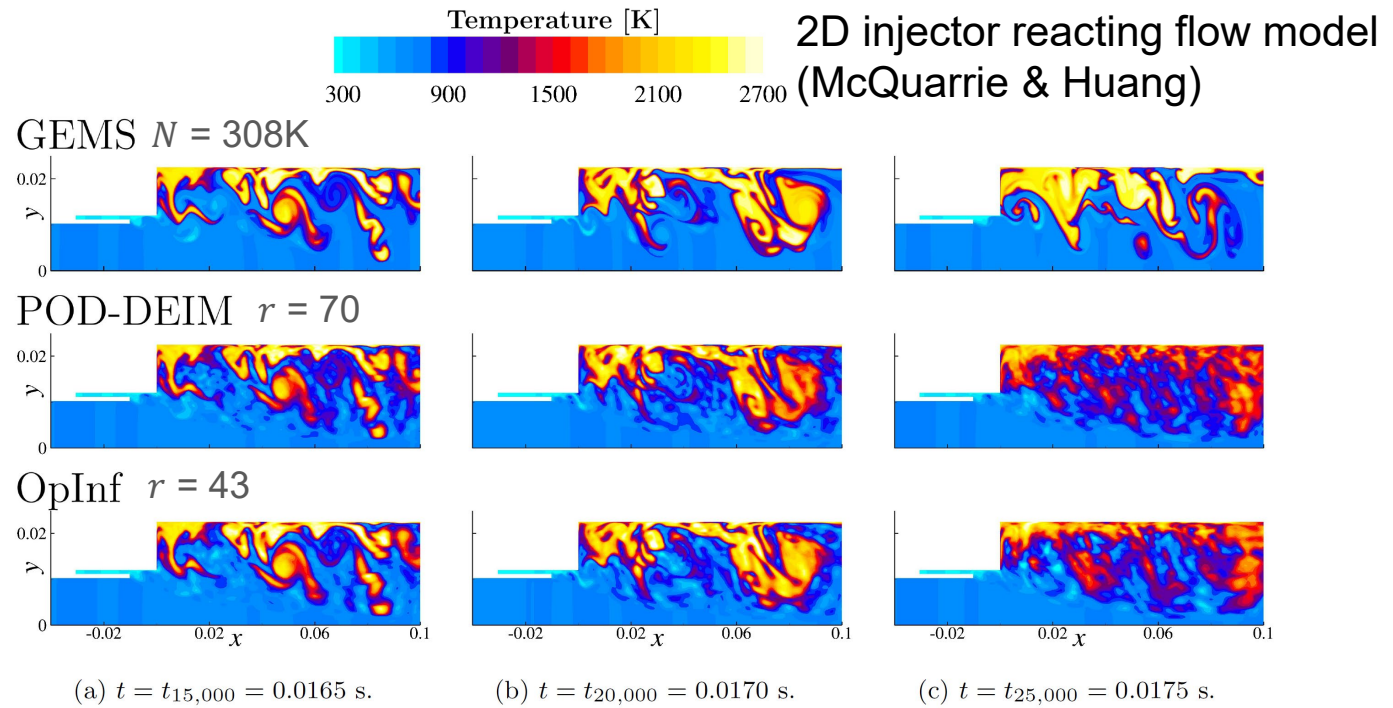
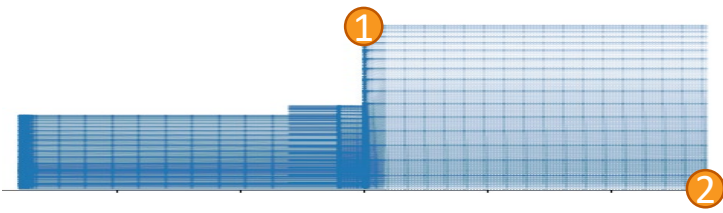
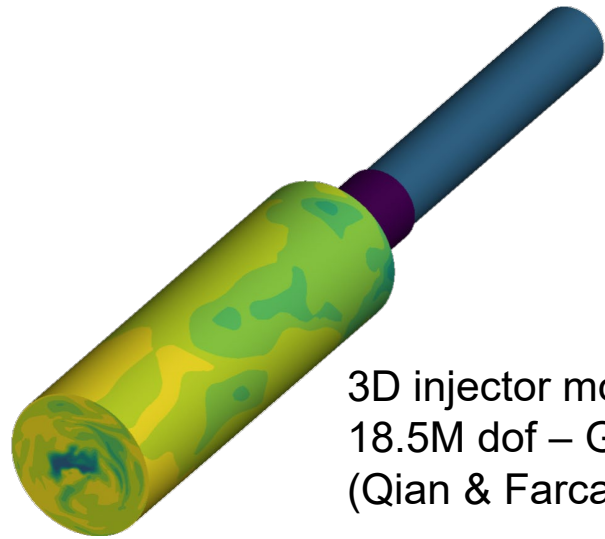


**Non-intrusive learning** by inferring reduced operators from simulation data [Peherstorfer & W., 2016]

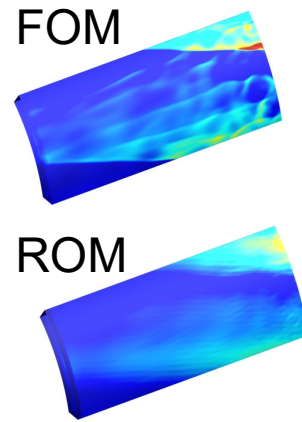
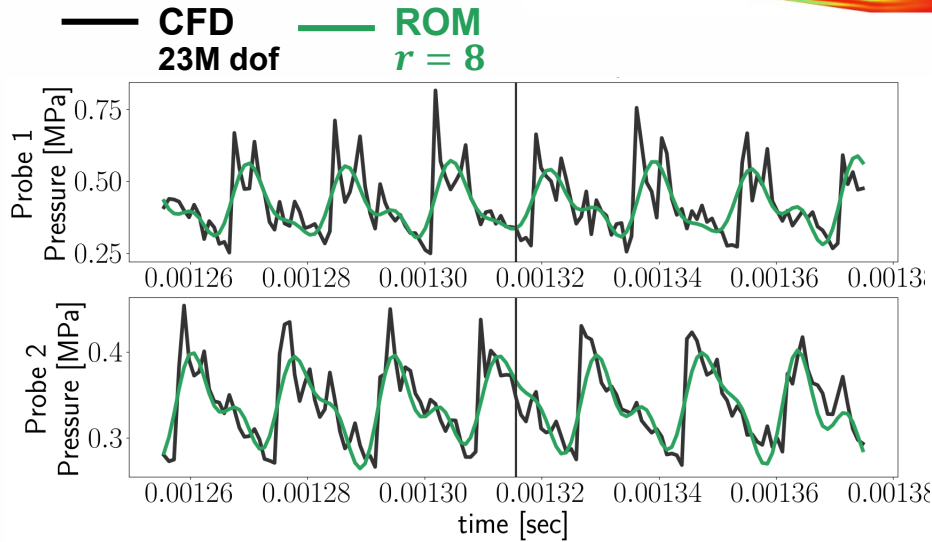
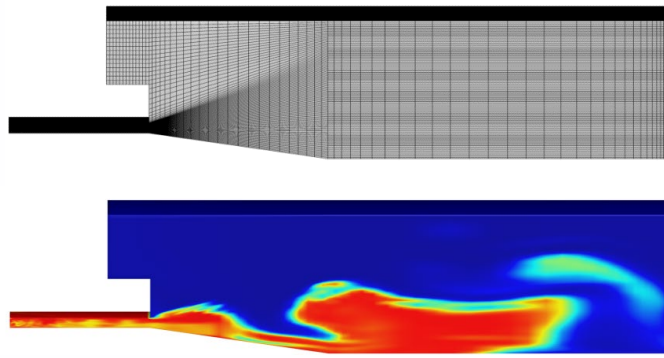
## **Operator Inference** is non-intrusive; requires only snapshot data

1. Generate snapshots from high-fidelity simulation
2. Compute POD basis (SVD) and snapshot low-dimensional representation
3. Solve linear least squares minimization problem to infer the low-dimensional model

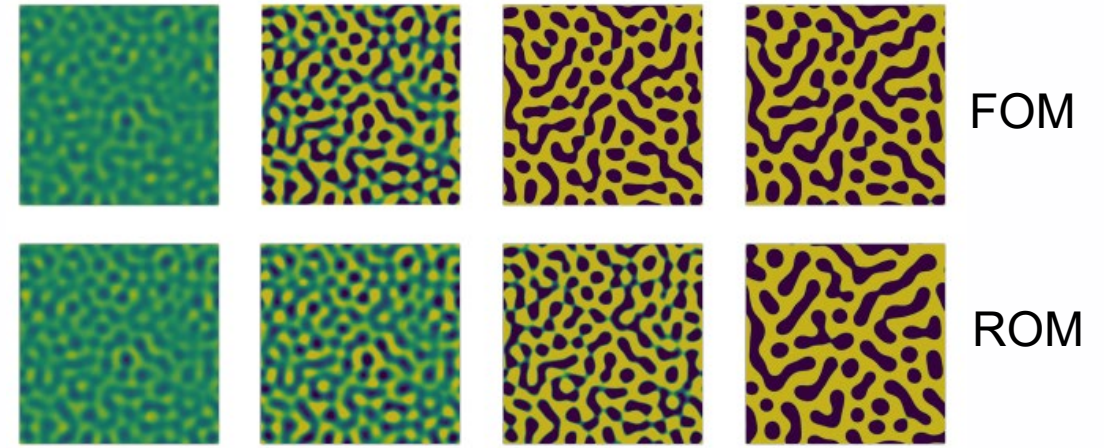
**Operator Inference ROMs are competitive in accuracy with state-of-the-art intrusive ROMs** but are much faster/simpler to implement and faster to solve



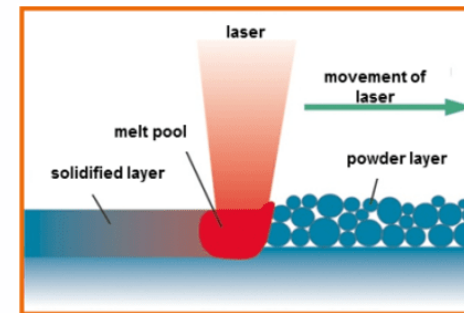
# Rotating Detonation Engine (Farcas)



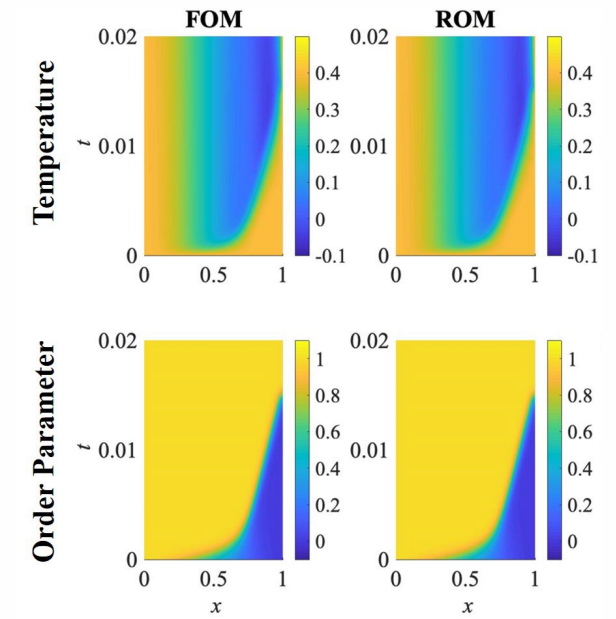
# Phase-field modeling (Geelen)



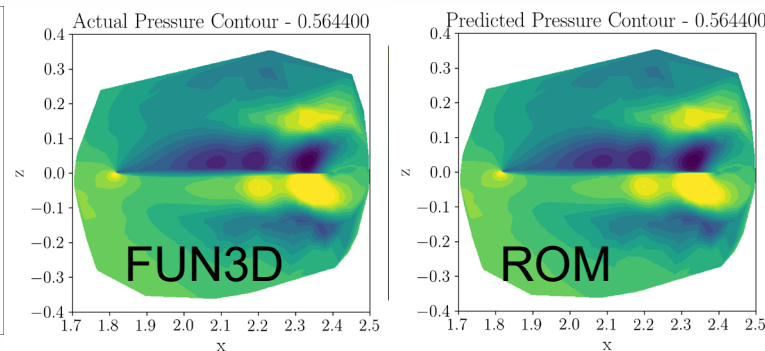
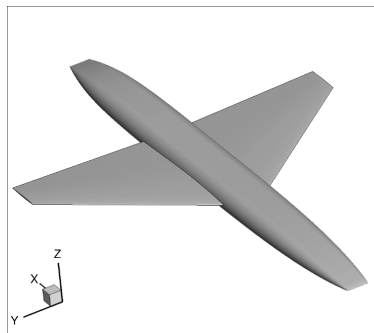
# Solidification in additive manufacturing (Khodabakhshi)



<https://www.bintoa.com/powder-bed-fusion>



# VAT Wing (Zastrow & Chaudhuri)



Structure + predictivity  
from physics

Non-intrusivity + flexibility  
from machine learning

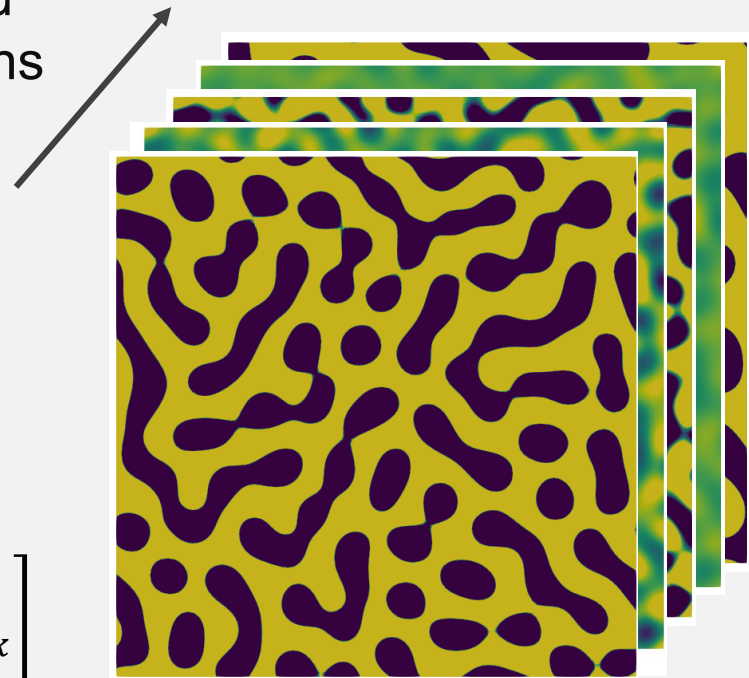
# Operator Inference

**What more can machine learning  
concepts bring to model reduction?**



# Why are many aerospace problems challenging for model reduction?

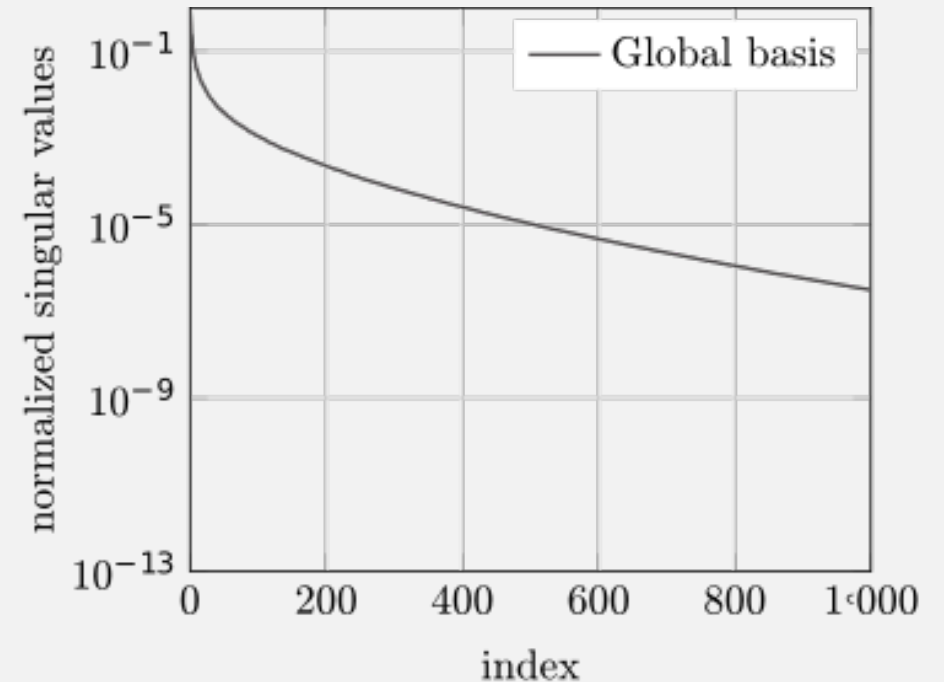
Snapshot collection  
across different  
time steps and  
initial conditions



Cahn-Hilliard  
phase-field model

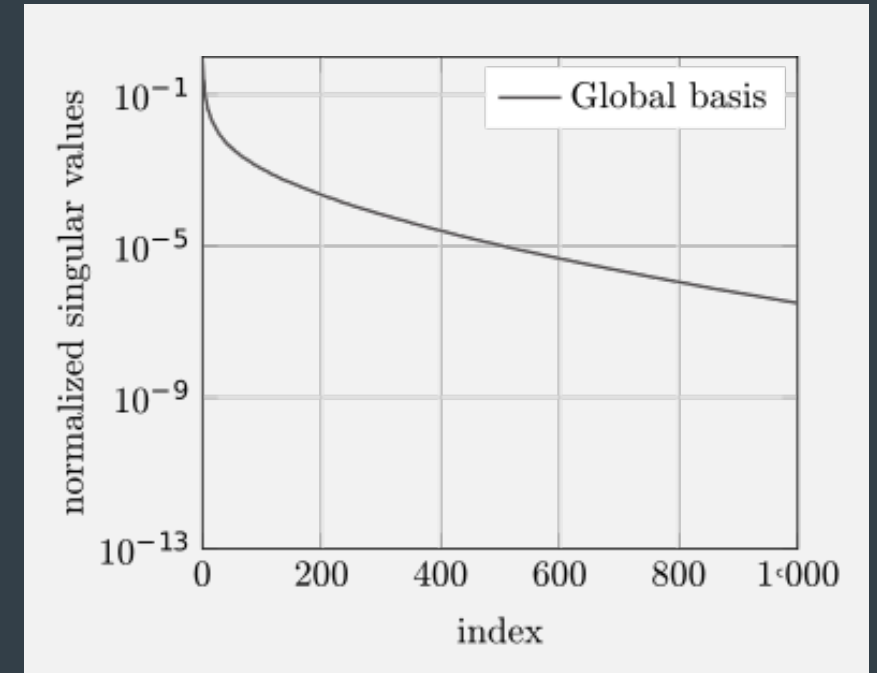
$$\mathbf{S} = \begin{bmatrix} | & | & \cdots & | \\ \mathbf{s}_0 & \mathbf{s}_1 & & \mathbf{s}_k \\ | & | & & | \end{bmatrix}$$

Slow decay of the singular values  
→ reduced model has high dimension



# Constructing reduced-order models is challenging for advection-dominated and multiscale problems

- Addressed by: adaptive model reduction, interpolation, nonlinear manifolds, dictionaries of ROMs, problem-specific registration, domain decomposition, ...  
[Amsallem, Beran, Farhat, Haasdonk, Ohlberger, Patera, Peherstorfer, Rozza, Ryckelynck, Stamm, Vega, Zahr, ...]
- These approaches are all intrusive, limiting their applicability



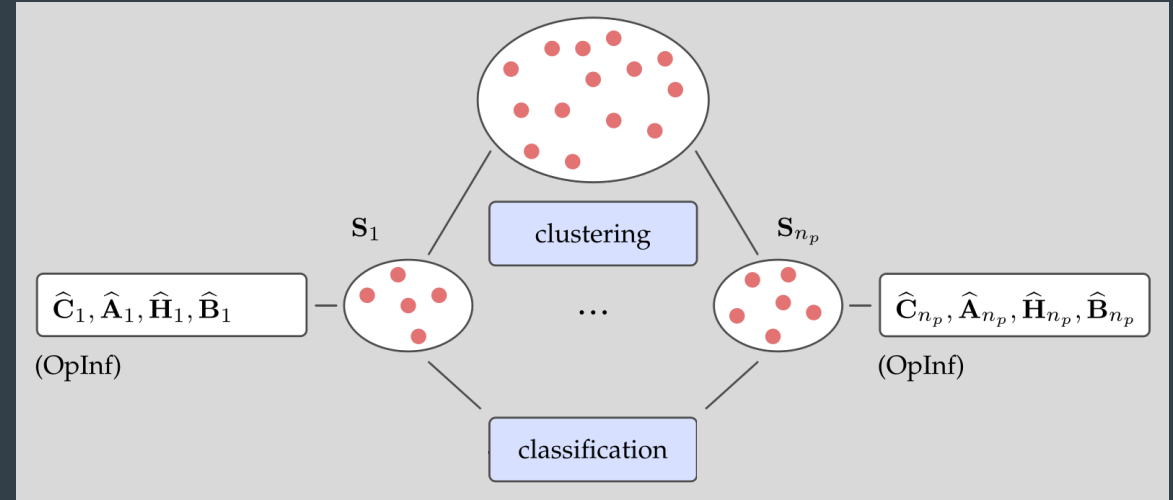
→ **Localized Operator Inference:**  
**non-intrusive physics-based (Operator Inference) +**  
**approximation power of dictionaries of localized ROMs**

# Localized Operator Inference – Divide & Conquer

[Geelen & W., Phil. Trans. Royal Society A, 2021]

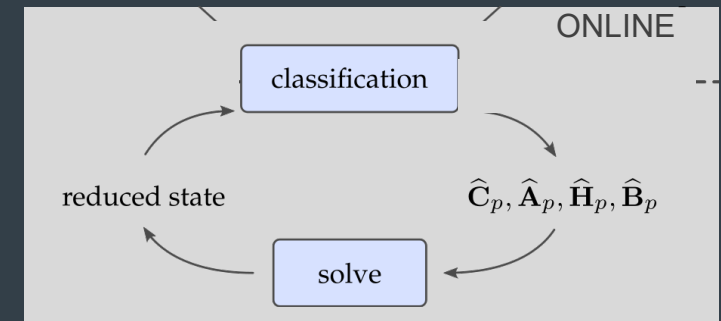
## Offline

1. **Data collection & clustering**  
using unsupervised learning methods
2. **Train a classifier**  
for selecting the local ROM
3. **Learn a set of cluster-specific ROMs**



## Online

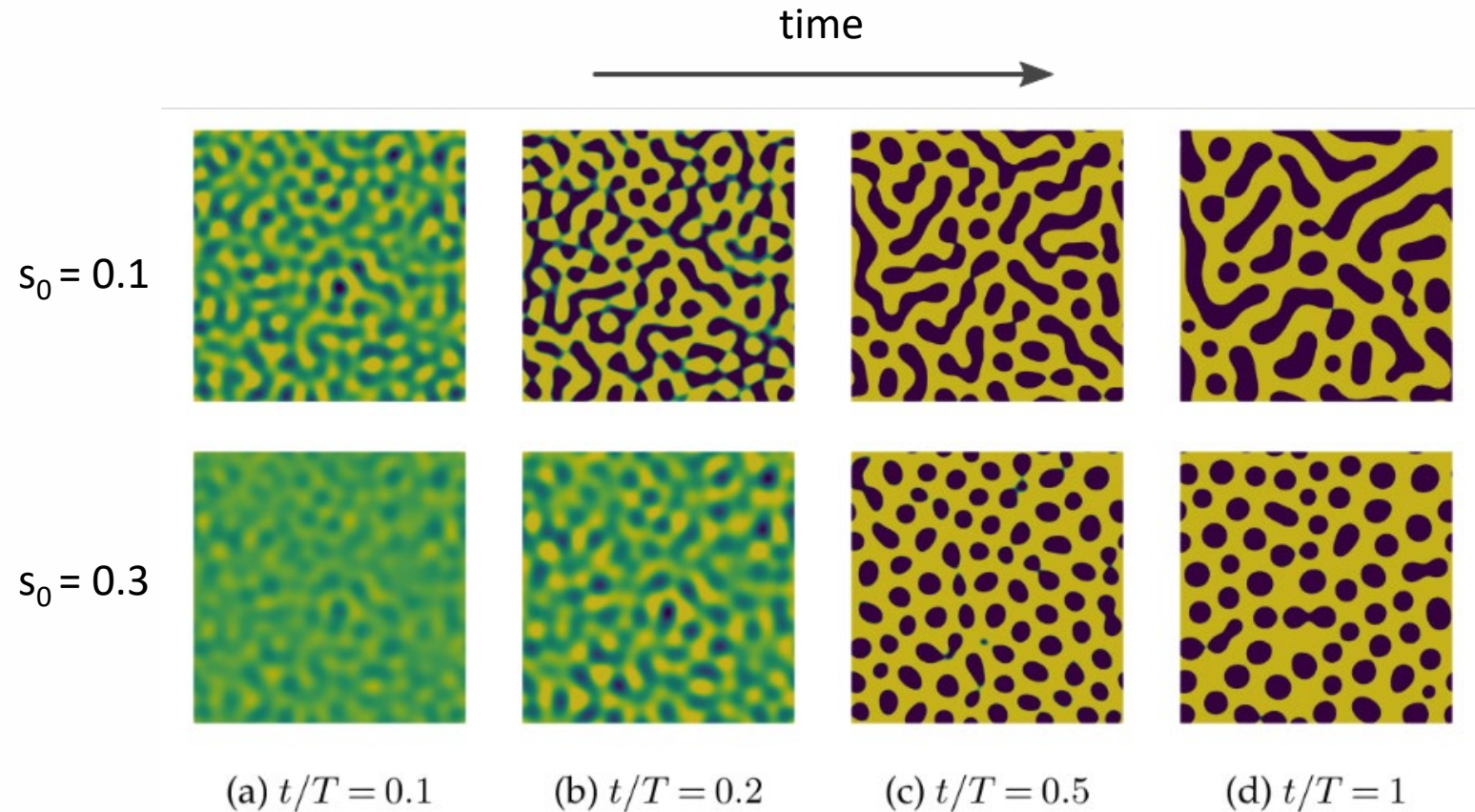
1. **Select ROM** – using the state as an indicator, select which local ROM to employ
2. **Solve** – evaluate the ROM using reduced model operators corresponding to the selected local ROM



# Reducing a Cahn-Hilliard phase-field model

$$\frac{\partial}{\partial t} s(\mathbf{x}, t) = M \nabla^2 \left( s^3(\mathbf{x}, t) - s(\mathbf{x}, t) - \ell \nabla^2 s(\mathbf{x}, t) \right)$$

Different initial conditions give rise to different kinematics across different **temporal** and **spatial** scales



# Localized Operator Inference: Offline Phase

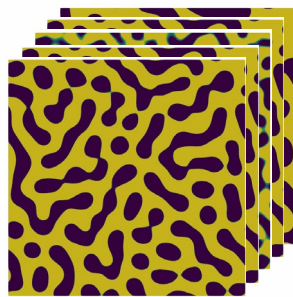
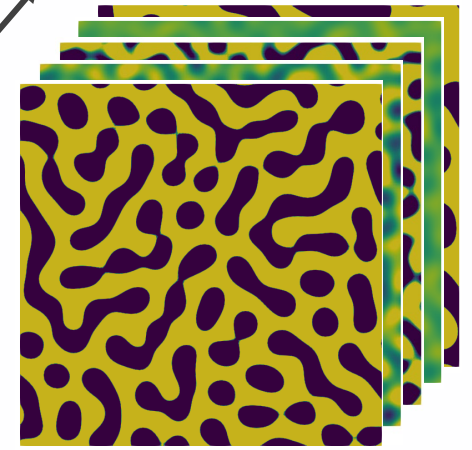
## Step 1 – Data collection & clustering

- Snapshot training data (e.g., from high-fidelity codes)

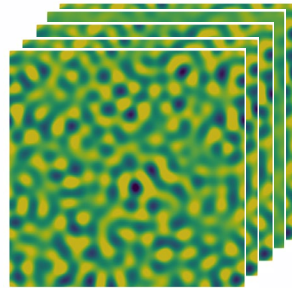
$$\mathbf{S} = \begin{bmatrix} | & | & \cdots & | \\ \mathbf{s}_0 & \mathbf{s}_1 & \cdots & \mathbf{s}_k \\ | & | & \cdots & | \end{bmatrix}, \dot{\mathbf{S}} = \begin{bmatrix} | & | & \cdots & | \\ \dot{\mathbf{s}}_0 & \dot{\mathbf{s}}_1 & \cdots & \dot{\mathbf{s}}_k \\ | & | & \cdots & | \end{bmatrix}, \mathbf{U} = \begin{bmatrix} | & | & \cdots & | \\ \mathbf{u}_0 & \mathbf{u}_1 & \cdots & \mathbf{u}_k \\ | & | & \cdots & | \end{bmatrix}$$

- Compute a global POD basis  $\bar{\mathbf{V}} \in \mathbb{R}^{N \times q}$   
(compression for clustering and classification)
- Low-dimensional data representation  $\tilde{\mathbf{S}} = \bar{\mathbf{V}}^T \mathbf{S}$ :
- Partition training data into  $n_p$  snapshot clusters using unsupervised learning

Snapshot collection  
across different  
timesteps and  
initial conditions

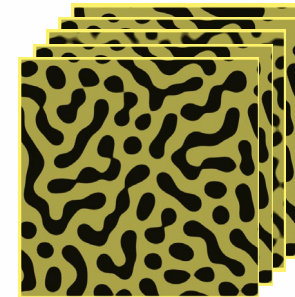


cluster 1



cluster 2

...



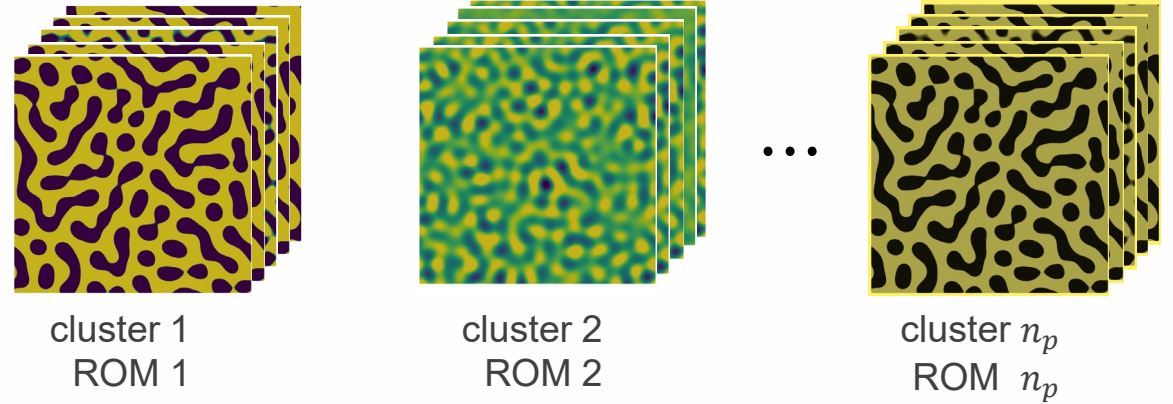
cluster  $n_p$

# Localized Operator Inference: Offline Phase

## Step 2 – Train the classifier

- Nearest neighbor classifier maps from low-dimensional state  $\tilde{\mathbf{S}}$  to cluster index

$$\text{classifier: } \tilde{\mathbf{s}} \rightarrow \{1, 2, \dots, n_p\}$$



## Step 3 – Learn $n_p$ cluster-specific Operator Inference ROMs

- Localized ROMs have cubic form (inherits structure of Cahn Hilliard):

$$\frac{d}{dt} \hat{\mathbf{s}}_p(t) = \hat{\mathbf{A}}_p \hat{\mathbf{s}}_p(t) + \hat{\mathbf{G}}_p (\hat{\mathbf{s}}_p(t) \otimes \hat{\mathbf{s}}_p(t) \otimes \hat{\mathbf{s}}_p(t))$$

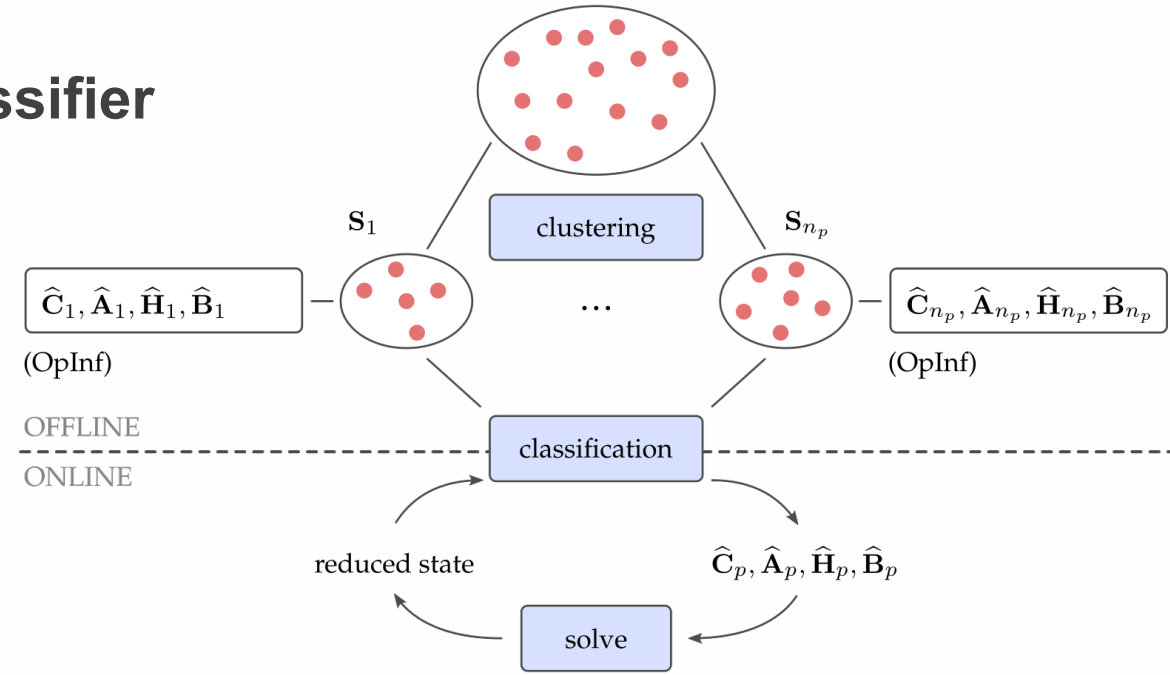
- Solve linear least squares to infer localized operators  $\hat{\mathbf{A}}_p$  and  $\hat{\mathbf{G}}_p$

$$\arg \min_{\hat{\mathbf{A}}_p, \hat{\mathbf{G}}_p} \left( \left\| \hat{\mathbf{A}}_p \hat{\mathbf{S}}_p + \hat{\mathbf{G}}_p (\hat{\mathbf{S}}_p \odot \hat{\mathbf{S}}_p \odot \hat{\mathbf{S}}_p) - \dot{\hat{\mathbf{S}}}_p \right\|_F^2 + \lambda_{1,p} \|\hat{\mathbf{A}}_p\|_F^2 + \lambda_{2,p} \|\hat{\mathbf{G}}_p\|_F^2 \right)$$

# Localized Operator Inference: Online Phase

## Step 1 – Compute indicator and evaluate classifier

$$\tilde{\mathbf{s}} = \bar{\mathbf{V}}^\top \mathbf{V}_p \hat{\mathbf{s}}_p \in \mathbb{R}^q \rightarrow \{1, 2, \dots, n_p\}$$



## Step 2 – ROM evaluation

- $p$ th ROM:

$$\frac{d}{dt} \hat{\mathbf{s}}_p(t) = \hat{\mathbf{A}}_p \hat{\mathbf{s}}_p(t) + \hat{\mathbf{G}}_p (\hat{\mathbf{s}}_p(t) \otimes \hat{\mathbf{s}}_p(t) \otimes \hat{\mathbf{s}}_p(t))$$

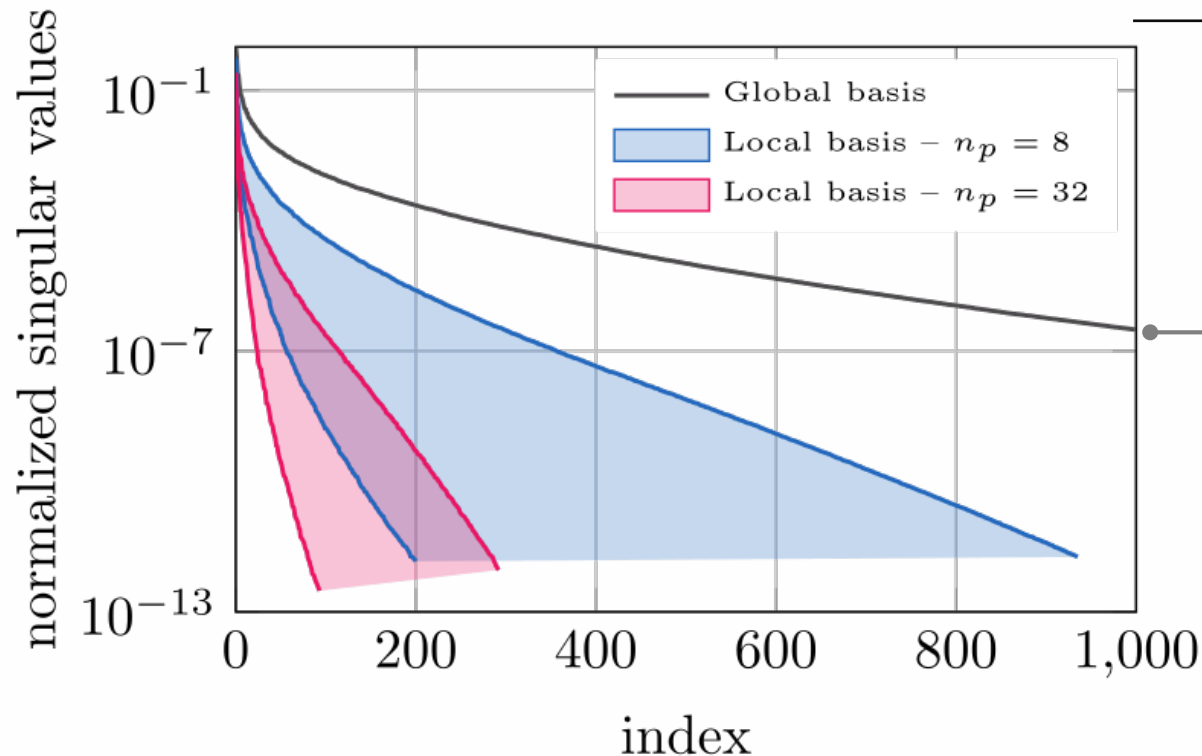
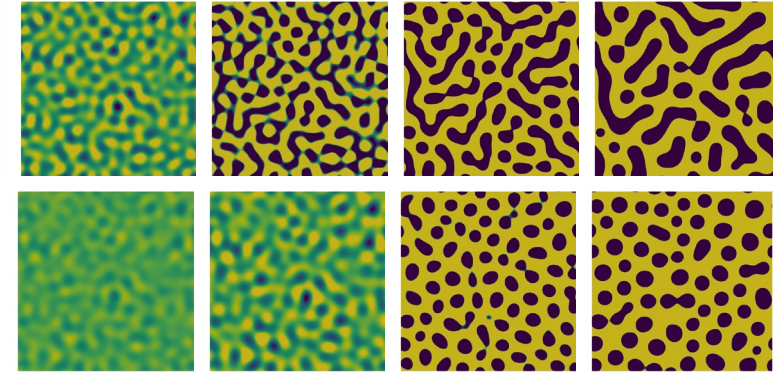
- When switching from cluster/ROM  $b$  to  $a$ , project the reduced state

$$\hat{\mathbf{s}}_a = \mathbf{V}_a^\top \mathbf{V}_b \hat{\mathbf{s}}_b \in \mathbb{R}^{r_a}$$

# Localization via clustering (unsupervised learning) is one way to mitigate slow singular value decay

$$\frac{\partial}{\partial t} s(\mathbf{x}, t) = M \nabla^2 \left( s^3(\mathbf{x}, t) - s(\mathbf{x}, t) - \ell \nabla^2 s(\mathbf{x}, t) \right)$$

snapshots seeded with different initial conditions



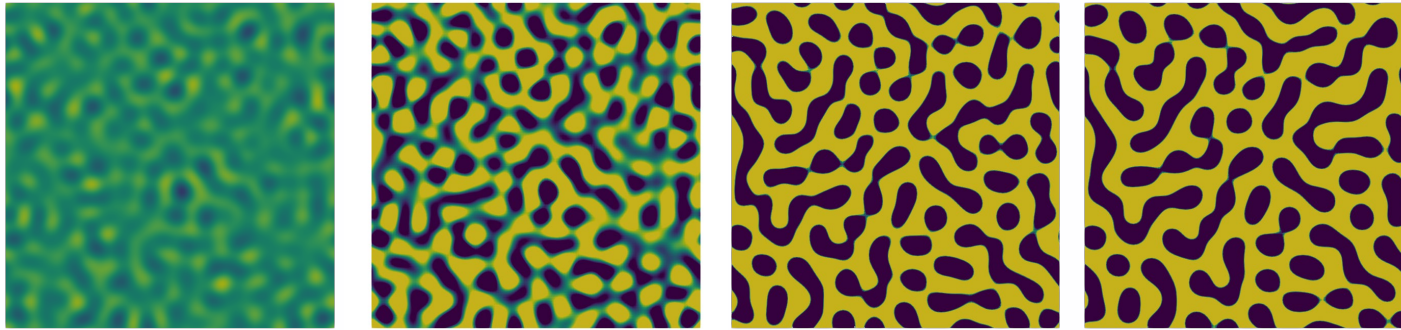
A global basis would require  $10^2$ – $10^3$  modes



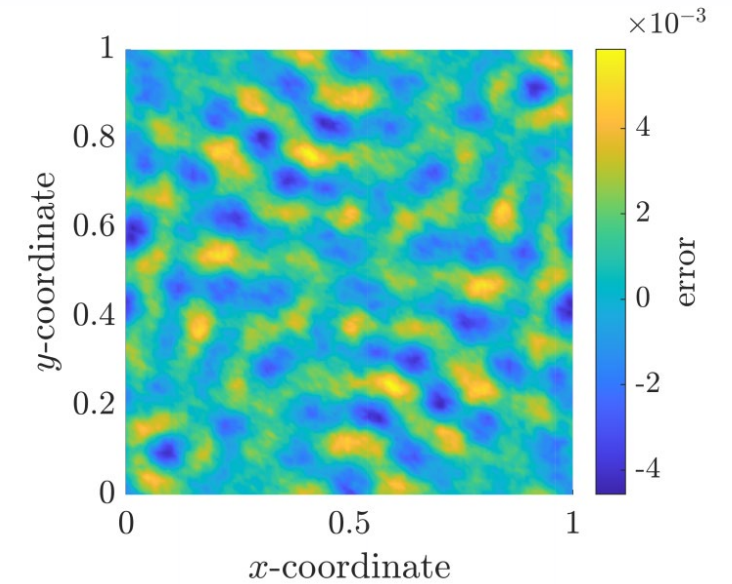
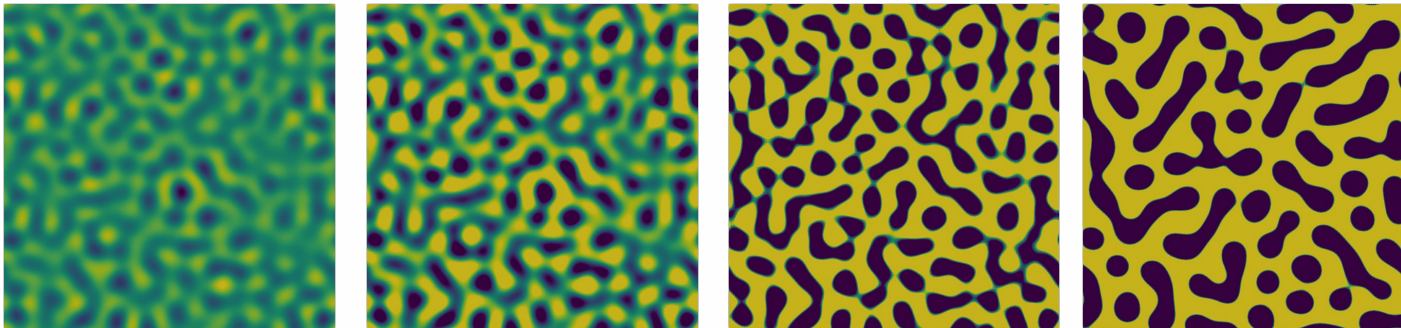
# Reduced-order model performance

FOM  $N = 16,384$

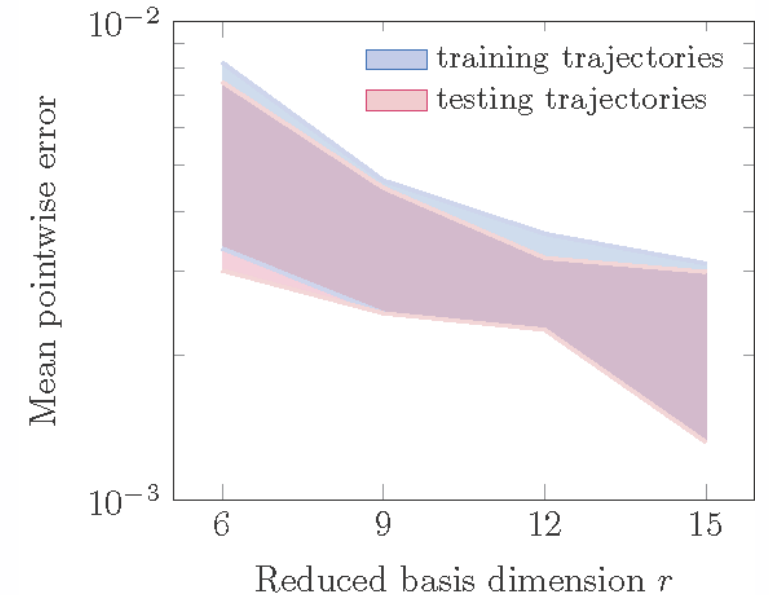
concentration  
5.0e-02 0.08 0.1 0.12 1.5e-01



Localized OpInf ROM with 32 snapshot clusters,  $r = 15$



**Error in the autocorrelations**



# COMPUTATIONAL SCIENCE enables AEROSPACE DESIGN

1

## Learning from data

The imperative of physics-based modeling & inverse theory

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2

## Reduced-order modeling

A critical enabler for accelerating predictive computations in support of engineering design

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3

## Operator Inference

Combining model reduction & machine learning to learn predictive reduced-order models

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4

## Outlook

**Machine Learning**

**Scientific Machine Learning**

**Reduced Order Modeling**

## **A range of tools for a range of use cases**

**Training data that cover the search space; need a fast efficient look-up table**

**Trusted expensive code base; black box approximations (response surfaces) are sufficient**

**Some training data but need to issue predictions beyond the training data**

**Trusted expensive code base; time and expertise to implement intrusive predictive reduced model**

# Challenges for **ENGINEERING DESIGN** in the age of **BIG DATA & BIG COMPUTE**

## **1 Predictive modeling for complex systems at scale**

Decisions demand a predictive window on the future

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## **2 Validation, verification & uncertainty quantification**

Achieving the levels of reliability and robustness needed for certified high-consequence decision-making

---

## **3 Data, models and decisions across multiple scales**

Scalable algorithms for calibration, data assimilation, optimization, uncertainty quantification, planning & control

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## **4 Optimal sensing strategies**

Digital twins, integrated sensor design, optimal experimental design (active learning), intelligent adaptive data acquisition

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## **5 Legacy codes and processes**

Equipping our processes and our people with state-of-the-art computational science & computer science

# Data-driven decisions

building the mathematical foundations and computational methods to enable design of the next generation of engineered systems

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